

2506/305
AIRCRAFT MECHANICAL
TECHNOLOGY II
Oct./Nov. 2019
Time: 3 hours



THE KENYA NATIONAL EXAMINATIONS COUNCIL
DIPLOMA IN AERONAUTICAL ENGINEERING
(AIRFRAMES AND ENGINES OPTION)
MODULE III
AIRCRAFT MECHANICAL TECHNOLOGY II
3 hours

INSTRUCTIONS TO CANDIDATES

You should have the following for this examination:

Answer booklet;

Drawing instruments;

Mathematical tables/Non-programmable scientific calculator;

Thermodynamic and Transport properties of fluid tables by Rogers and Mayhew.

This paper consists of EIGHT questions in TWO sections; A and B.

Answer THREE questions from section A and TWO questions from section B.

All questions carry equal marks.

Maximum marks for each part of a question are as shown.

Candidates should answer the questions in English.

This paper consists of 6 printed pages.

Candidates should check the question paper to ascertain that all the pages are printed as indicated and that no questions are missing.

SECTION A: THERMODYNAMICS

Answer **THREE** questions from this section.

1. (a) Define the following terms as applied to thermodynamics:

- (i) system;
- (ii) boundary.

(2 marks)

(b) (i) Distinguish between isothermal and adiabatic processes.

(ii) State the first law of thermodynamics.

(4 marks)

(c) (i) Show that for a perfect gas:

$$C_p = \frac{\gamma R}{(\gamma - 1)}$$

where C_p = specific heat capacity at constant pressure;
 γ = ratio of specific heat capacity at constant pressure to the
specific heat capacity at constant volume;
 R = gas constant.

(ii) A quantity of carbon dioxide is compressed from an initial state of 0.092 m³, 1.2 bar to a final state of 0.043 m³ and 4.2 bar. The observed temperature rise is 154 K. Determine the:

- (I) gas constant;
- (II) mass of gas present;
- (III) increase in internal energy of the gas.

Assume carbon dioxide to be a perfect gas.

Take: $C_p = 0.82$ kJ/kg K and $C_v = 0.63$ kJ/kg K

(14 marks)

2. (a) With the aid of a P-V diagram, describe the processes of the ideal dual combustion cycle. (8 marks)

(b) Show that for an air standard Otto cycle:

$$\frac{T_2}{T_1} = r_v^{\gamma-1}$$

Where:

- T_2 = temperature at the end of compression;
 T_1 = temperature at the beginning of compression;
 r_v = compression ratio;
 γ = adiabatic index.

(6 marks)

(c) In an air standard Otto cycle, the maximum and minimum temperatures are 1,500 °C and 14 °C. The heat supplied per kg of air is 700 kJ. Determine the:

- (i) compression ratio;
(ii) thermal efficiency.

Take $\gamma = 1.4$; $C_v = 0.718$ kJ/kg K.

(6 marks)

3. (a) Define each of the following as used in combustion of fuels:

- (i) stoichiometric mixture;
(ii) ultimate analysis;
(iii) mixture strength;
(iv) calorific value.

(4 marks)

(b) A petrol engine uses ethyl alcohol (C_2H_6O) for combustion. Determine the:

- (i) stoichiometric air fuel ratio by mass;
(ii) wet and dry analysis by volume of the exhaust gas in b(i);
(iii) air fuel ratio for a mixture strength of 80%.

(16 marks)

4. (a) With the aid of a T-S diagram, explain the function of an intercooler in a gas turbine plant. (4 marks)

$$Q = P \quad Q = \dot{Q} = V = \frac{Q}{A}$$

- (b) In an aircraft gas turbine unit, a high pressure stage turbine drives the compressor and a low pressure stage turbine drives the propeller through suitable gearings. The overall pressure ratio is 5/1 and the maximum temperature is 700° C. The isentropic efficiencies of the compressor, high pressure turbine and low pressure turbine are 0.8, 0.82 and 0.86 respectively. The air intake conditions are 1.013 bar and 30° C.

- (i) Draw the plant and T-S diagrams:
(ii) Calculate the:
(I) pressure between turbine stages;
(II) thermal efficiency.

Take: $CP_a = 1.005 \text{ kJ/kg K}$, $CP_g = 1.15 \text{ kJ/kg K}$;
 $\gamma = 1.4$ for both compression and expansion.

(16 marks)

5. (a) Distinguish between the following types of heat exchangers:

- (i) regenerative and evaporative;
(ii) parallel flow and counter flow.

(4 marks)

- (b) (i) State the Fourier's law of conduction.

- (ii) Show that for a single cylindrical wall of unit length, the logarithmic mean area A_m is given by:

$$A_m = \frac{2\pi(r_2 - r_1)}{\ln\left(\frac{r_2}{r_1}\right)},$$

where: r_1 = internal radius;
 r_2 = external radius.

(9 marks)

- (c) In a parallel flow tubular heat exchanger, hot gases are cooled from 600° C to 200° C by water initially at 18° C. The mass flow rate of the gas is 0.4 kg/s and that of water 0.5 kg/s. The specific heat capacities of the gas and water are 1.2 kJ/kg K and 4.2 kJ/kg K respectively. The overall heat transfer coefficient from gases to water is 150 W/m²K.

- (i) Sketch the temperature variation along the length of the heat exchanger.
(ii) Calculate the surface area required.

(7 marks)

SECTION B: FLUID MECHANICS

Answer TWO questions from this section.

6. (a) Assuming that the frictional resistance per unit area of pipe wall q is proportional to the square of the mean velocity of flow, show that the loss of head due to friction h_f in a pipeline is given by the formula:

$$h_f = \frac{4fLV^2}{2dg}$$

Where: f = resistance coefficient;
 L = length of pipe;
 V = velocity;
 d = diameter of pipe;
 g = gravitational acceleration.

(9 marks)

- (b) The discharge through a horizontal pipe of varying cross-section is $0.96 \text{ m}^3/\text{s}$. The first part of the pipe is 3 m long and 0.2 m diameter. The second part is 4 m long and 0.5 m diameter. The third part is 5 m long and 0.3 m diameter. The coefficient of contraction $C_c = 0.6$ and frictional coefficient $f = 0.01$.

If the change of sections are sharp, determine the total loss of head due to:

- (i) sudden enlargement;
 (ii) sudden contraction;
 (iii) friction.

$$h_e = \frac{V^2}{2g} \left[1 - \frac{A_1}{A_2} \right]^2$$

(11 marks)

$$h_c = \frac{V^2}{2g} \left[\frac{1}{C} - 1 \right]^2$$

7. (a) Define the following terms as used in fluid mechanics:

- (i) coefficient of dynamic viscosity;
 (ii) kinematic viscosity.

(3 marks)

- (b) Show that the total discharge Q of a viscous incompressible fluid flowing between parallel flat plates is given by:

$$Q = \frac{Pbh^3}{12\mu L}$$

where: P = pressure difference between the ends of the plates;
 μ = coefficient of dynamic viscosity;
 L = length of plates;
 h = distance between the plates;
 b = width of the plates.

(13 marks)

- (c) A dashpot consists of a cylinder 60 mm diameter in which slides a piston 90 mm long. The radial clearance between the piston and the cylinder is 1.5 mm. The cylinder is filled with an oil of dynamic viscosity 0.1 kg/ms. If the piston is pushed by a force of 200 N, determine the volume of oil leaving the cylinder per second. (4 marks)

8. (a) Define the following terms as applied to pumps:

- (i) specific speed;
(ii) coefficient of discharge.

(2 marks)

- (b) (i) State two assumptions made while deriving the equation for work done by a centrifugal pump.
(ii) With the aid of a diagram, show that for a centrifugal pump, the work done per unit weight is given by the equation:

$$\text{Work done per unit weight} = \frac{W_2 u_2}{g}$$

Where: w_2 = tangential component of absolute velocity at exit
(whirl velocity at exit);
 u_2 = tangential velocity of impellor at outlet;
 g = gravitational acceleration.

(10 marks)

- (c) A centrifugal pump running at 900 rev/min works against a total head of 24 m. The external diameter of the impeller and outlet width are 520 mm and 80 mm respectively. The vanes angle at outlet is 36° and the manometric efficiency is 80%. Determine the:

- (i) flow velocity at outlet;
(ii) absolute velocity of the fluid leaving the vane;
(iii) angle of absolute velocity at exit;
(iv) impeller power.

(8 marks)

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